

**III B.Tech I Semester Supplementary Examinations, May/Jun 2009
AIRCRAFT STRUCTURES-I
(Aeronautical Engineering)**

Time: 3 hours

Max Marks: 80

**Answer any FIVE Questions
All Questions carry equal marks**

1. Explain in detail the boundary conditions for various kinds of beams with neat sketches in terms of slopes and deflections. [16]
2. A cantilever beam of span L is subjected to uniform varying load 'w' per unit width, varying from zero at the free end to maximum at the fixed end followed by a clock wise couple, M at the free end. Find the slope and deflection at the free end using method of super position, assuming uniform flexural rigidity, EI. [16]
3. A statically indeterminate system is shown in (figure3) for a propped cantilever. Find the prop reaction at C. Find the net downward deflection [16]
 $E=2 \times 10^5 \text{ N/mm}^2$
 $I=1000 \times 10^4 \text{ mm}^4$

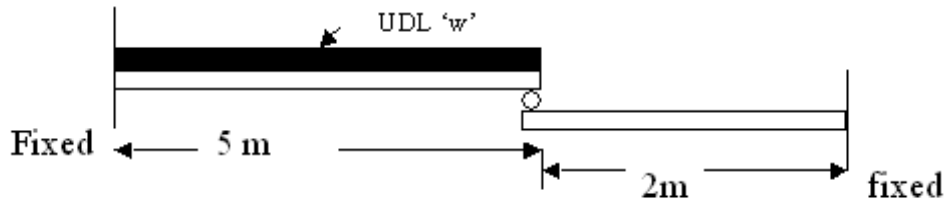


Figure 3

4. A frame is loaded as shown in (figure4) at point D. Determine the vertical deflection using Strain energy method. [16]

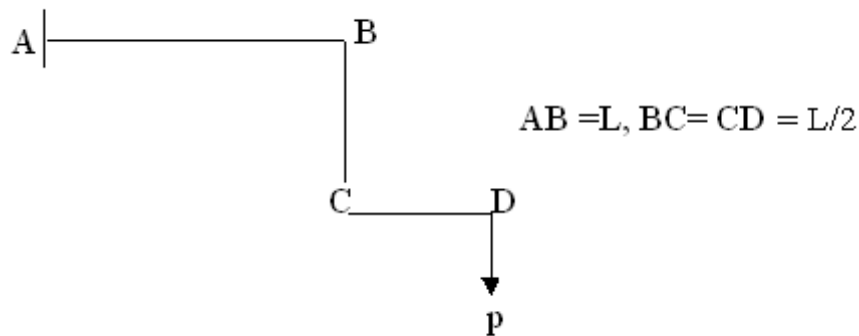


Figure 4

5. Find the deflection and slope using unit load method at the free end of a cantilever beam loaded as shown in figure5, assuming uniform flexural rigidity, EI. [16]

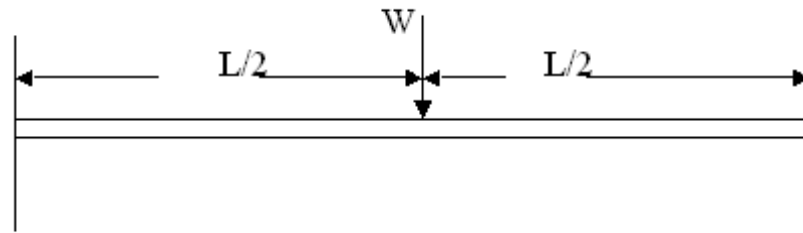


Figure 5

6. A column having a T-section with a flange 120mmX16mm and web 150X16mm is 3 m long. Assuming the column to be hinged at both ends. Find the crippling load by Euler's formula.
 $E=2 \times 10^5 \text{ N/mm}^2$ [16]
7. Compare the crippling loads given by Rankine's formula and Euler's Formula for a tubular strut having outer and inner diameters as 37.5 mm And 32.5 mm respectively loaded through pin joints at both ends. The critical stress is 32.5 KN/cm^2 .
 $k_1 = 1/7500$ $E = 2 \times 10^5 \text{ N/mm}^2$. [16]
8. A beam-column of span L , which fixed at both the ends is subjected to axial compressive load P at the ends followed by a transverse UDL 'w' per unit length over the entire span
- (a) Write the beam column equations considering buckling at a section
- (b) Give the boundary conditions. [16]

III B.Tech I Semester Supplementary Examinations, May/Jun 2009
AIRCRAFT STRUCTURES-I
(Aeronautical Engineering)

Time: 3 hours

Max Marks: 80

Answer any FIVE Questions
 All Questions carry equal marks

1. Two cantilever beams AB and CD of equal length 4000 mm of equal flexural rigidity $EI=24 \times 10^{12} \text{ Nmm}^2$ are fixed to a wall (AB above CD) in vertical plane horizontally and are inter connected by a steel rod BD ($E=200\text{Gpa}$) to the free ends B and D. Rod BD is 5000mm long and has a cross-sectional area of 300mm^2 . Find the deflection of cantilever AB at B due to a downward force of $P=50 \text{ KN}$ applied at D considering the elongation of the rod. [16]
2. A simple supported flitched beam (composite) of span 2m carries a U.D.L of 10 KN/m over its entire span. The beam is made of aluminum $100 \times 50\text{mm}_2$ reinforced with steel plates $50 \times 10 \text{ mm}^2$. Determine the maximum deflection and slope for the composite beam. $E_{steel} = 3.5 \times E_{wood}$. $E_{steel} = 2 \times 10^5 \text{ N/mm}^2$. [16]
3. A SSB of span L is carrying a UDL, 'w' over its entire length and propped at the center. Find the prop reaction and draw the SFD and BMD. [16]
4. Determine the deflection under 60 KN load for the beam ACB shown in figure4 using Castiglino's theorem $E=2 \times 10^5 \text{ N/mm}^2$ [16]

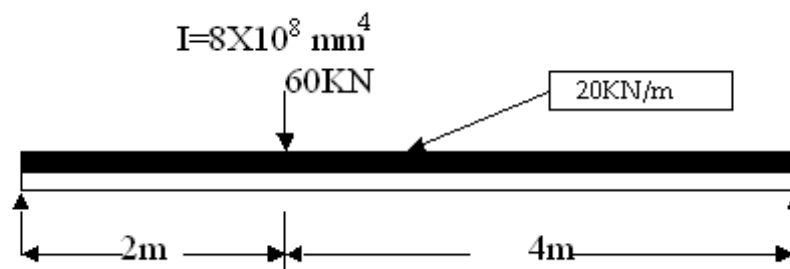


Figure 4

5. Find the forces in all the members of the structure and tabulate the results as shown in figure below 5. [16]

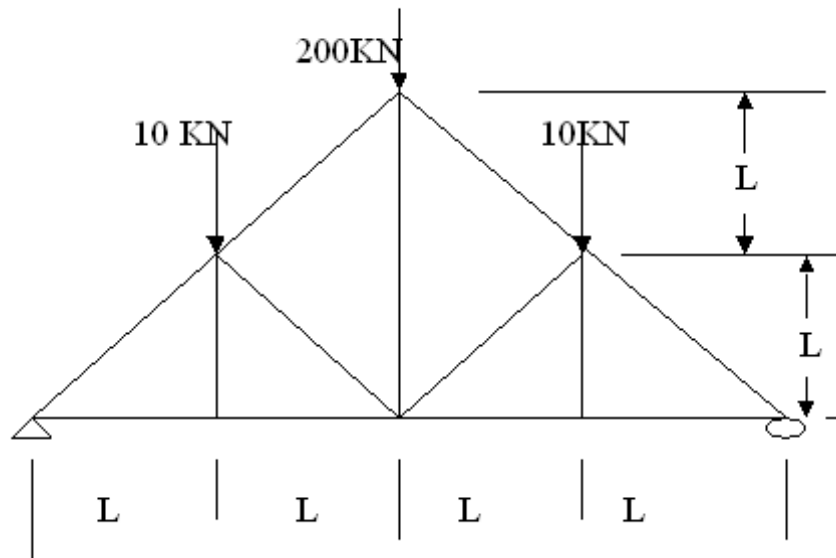


Figure 5

6. Derive Euler critical load formula for a column with one end fixed ,other end hinged. [16]
7. A cast iron of hollow circular section 200mm external dia ,thickness of metal 20mm,length 5m,has to take a load of 120 kN at an eccentricity of 20 mm from the geometrical axis. If the ends are fixed, calculate max. and min. stresses induced in the section taking $E=120 \text{ KN/mm}^2$.Also calculate max. permissible eccentricity so that no tension is induced anywhere in the section. [16]
8. A beam column of span L , which is fixed at both ends is subjected to moment M , at a distance $L/3$ from right end, followed by axial compressive load P ,
- (a) Write the beam column equations for buckling at a section
- (b) Give the boundary conditions. [16]

III B.Tech I Semester Supplementary Examinations, May/Jun 2009
AIRCRAFT STRUCTURES-I
(Aeronautical Engineering)

Time: 3 hours

Max Marks: 80

Answer any FIVE Questions
 All Questions carry equal marks

1. A cantilever beam of 20 cm wide and 40 cm thick with 5m is loaded with a U.D.L of 800N/m extending over a span of 3m from the free end Find the deflection at the free end using double integration method.
 $I = 16000 \times 10^4 \text{mm}^4$ $E = 2 \times 10^5 \text{N/mm}^2$ [16]
2. A cantilever beam of span 4 m is loaded with a u.d.l. of 50 KN/m over its entire span and a point load of 10 KN at its free end. Determine the maximum deflection and slope using superposition method.
 $I = 1.4 \times 10^7 \text{mm}^4$. $E = 2 \times 10^5 \text{N/mm}^2$ [16]
3. A cantilever of span 4m carries a point load 10KN at its free end .It is propped at 1m from the free end. Find the prop reaction at the support. [16]
4. State Castiglino's theorem for deflection of beams. Derive Castiglino's theorem. [16]
5. Find the forces in all the members of the structure and tabulate the results as shown in figure below 5. [16]

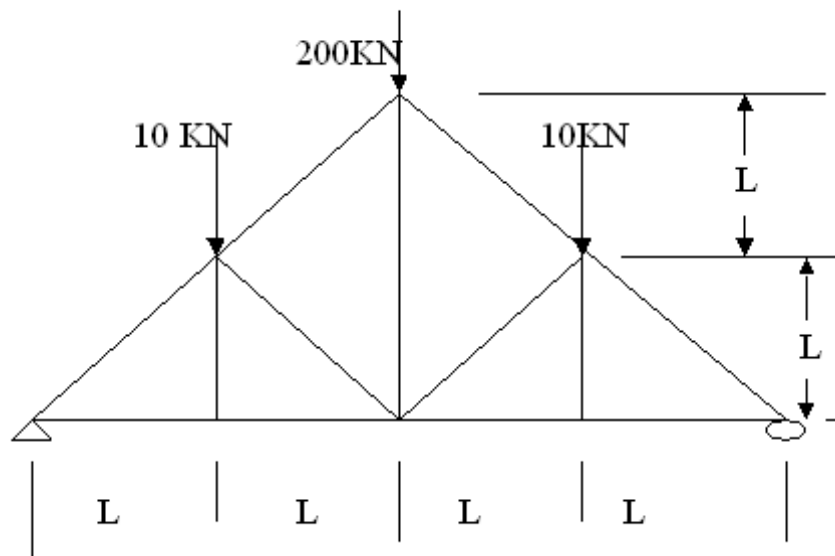


Figure 5

6. (a) Define the following terms: Columns, Buckling, Crippling load.
 (b) What are the assumptions made in Euler's theory of columns? [16]

7. A short cast iron column is tubular in section having 250 mm external Diameter and 200 internal diameter.
The column carries an eccentric load of 250 KN at a point distant 100 from its axis. Determine the maximum tensile and compressive stresses. Also draw the stress distribution. [16]
8. A beam-column of span L , which is hinged at one end and roller supported at the other end is subjected to axial compressive load P , at the ends followed by a clockwise couple M at the center
- (a) Write the beam column equations considering buckling at a section
(b) Give the boundary conditions. [16]

III B.Tech I Semester Supplementary Examinations, May/Jun 2009
AIRCRAFT STRUCTURES-I
 (Aeronautical Engineering)

Time: 3 hours

Max Marks: 80

Answer any FIVE Questions
 All Questions carry equal marks

- Two cantilever beams AB and CD of equal length 4000 mm of equal flexural rigidity $EI=24 \times 10^{12} \text{ Nmm}^2$ are fixed to a wall (AB above CD) in vertical plane horizontally and are inter connected by a steel rod BD ($E=200\text{Gpa}$) to the free ends B and D. Rod BD is 5000mm long and has a cross-sectional area of 300mm^2 . Find the deflection of cantilever AB at B due to a downward force of $P=50 \text{ KN}$ applied at D considering the elongation of the rod. [16]
- A cantilever beam of span 4 m is subjected to UDL of 20 KN/m over a span of 2m from the free end and a clock wise couple of 2 KN-m at the free end . Find the slope and deflection at the free end using the method of superposition. $EI = 3 \times 10^{14} \text{ N} - \text{mm}^2$ [16]
- A continuous beam with one end fixed is supported at B and C and loaded as shown in figure3. Find the end reactions as well as moments. $EI = 2 \times 10^{13} \text{ N/mm}^2$ [16]



Figure 3

- $AB = BC = 3\text{m}$ each AB is UDL of 3KN/m . At B point load of 60KN A cantilever beam is loaded as shown in figure4. Find the deflection at B using Castiglino's theorem. $EI = 2 \times 10^{14} \text{ Nmm}^2$ [16]

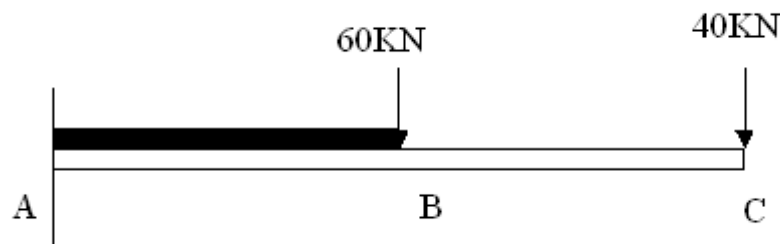
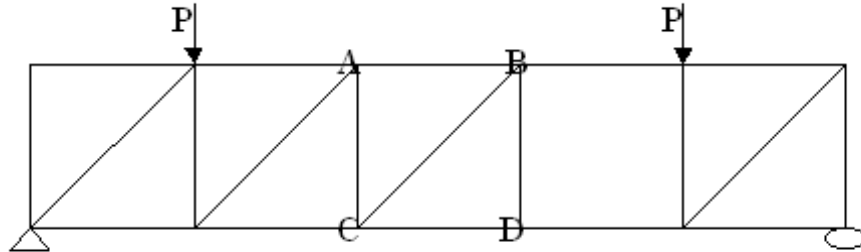


Figure 4

5. All the horizontal and vertical members are of equal length, L . A truss is loaded as shown in figure 5. Find the forces in the members AB, BC, CD and their nature and tabulate the forces. [16]



[

Figure 5

6. (a) Define the following terms: Columns, Buckling, Crippling load.
 (b) What are the assumptions made in Euler's theory of columns? [16]
7. A cast iron of hollow circular section 200mm external dia, thickness of metal 20mm, length 5m, has to take a load of 120 KN at an eccentricity of 20 mm from the geometrical axis. If the ends are fixed, calculate max. and min. stresses induced in the section taking $E=120 \text{ KN/mm}^2$. Also calculate max. permissible eccentricity so that no tension is induced anywhere in the section. [16]
8. A beam column of span L , which is fixed at both ends, is subjected to uniform varying transversely 'w' per unit width and axial compressive load, P at the ends
 (a) Write the beam column equation for buckling at a section
 (b) Give the boundary conditions. [16]
